

## Performance of the DAMPE silicon-tungsten tracker-converter during the first 5 years of in-orbit operations

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Since its launch, in December 2015, the satellite-based DAMPE (DARk Matter Particle Explorer) particle detector is taking data smoothly. The Silicon-Tungsten tracker-converter (STK) of DAMPE consists of six tracking planes (6x, 6y) of single-sided silicon micro-strip detectors mounted on seven support trays. The STK is able to measure the charge and precisely reconstruct the track of traversing charged particles. Tungsten plates (1 mm thick) are integrated in the second, third and fourth tray from the top to serve as  $\gamma \rightarrow e^+e^-$  converters. Commissioned rapidly after the launch, the STK is running extremely well since then. The STK in-orbit calibration and performance during its first more than 5 years of operation, including the noise behaviour and the thermal and mechanical stability, are presented in this contribution.

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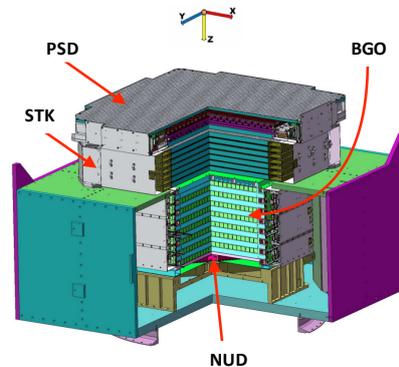
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## 1. Introduction

The DArk Matter Particle Explorer (DAMPE) is a satellite-based experiment for the detection of charged cosmic rays and gamma rays. It was launched on December 17, 2015 from the Jiuquan Satellite Launch Center in the Gobi Desert. DAMPE operates in a Sun-synchronous orbit around Earth at an altitude of about 500 km, with a period of about 95 minutes and an inclination of around  $97^\circ$ . Thanks to its four subdetectors, DAMPE detects electrons, positrons and photons from few GeV to 10 TeV, as well as protons and heavier nuclei from 10 GeV to 100 TeV, with excellent energy and angular resolutions [1]. The DAMPE subdetectors are (Fig. 1):

1. the Plastic Scintillator Detector (PSD), which measures the absolute value of the charge ( $|Z|$ ) of cosmic rays and acts as an anti-coincidence system for charged particles in the gamma-ray detection;
2. the Silicon-Tungsten tracKer-converter (STK), which is described in detail in the next section;
3. the imaging calorimeter made of 14 layers of Bismuth Germanium Oxide (BGO) crystals in a hodoscopic arrangement with a total thickness of about 32 radiation lengths, which measures the energy of the incoming particles and can distinguish electrons from hadrons;
4. the NeUtron Detector (NUD), a boron-doped plastic scintillator detecting delayed neutrons coming from the hadronic interactions at high energies, which boosts the electron/hadron separation power.

DAMPE is taking data smoothly since its launch, except for two days in May 2021, recording about five million events per day.



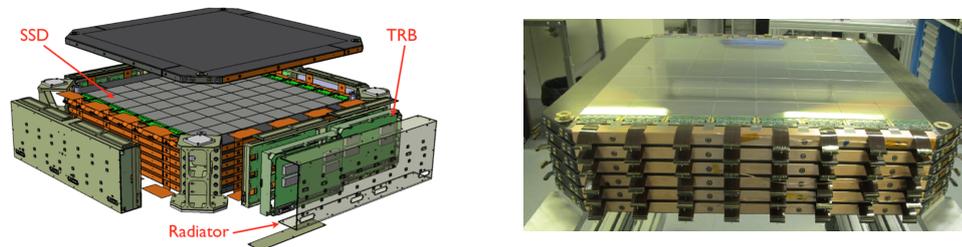
**Figure 1:** Schematic view of the DAMPE detector. From the top, the 4 subdetectors are: the plastic scintillator detector (PSD), the silicon-tungsten tracker-converter (STK), the bismuth germanate calorimeter (BGO) and the neutron detector (NUD).

## 2. The Silicon-Tungsten tracKer-converter (STK)

The DAMPE Silicon-Tungsten tracKer-converter (STK) [2] was designed for the purpose of reconstructing the trajectories and the charge absolute value ( $|Z|$ ) of charged cosmic rays,

favour the conversion of gamma-rays and reconstruct the tracks of the generated electrons and positrons. The STK was rapidly commissioned after the launch and the data acquisition started on December 30, 2015, less than two weeks after the launch. The STK has been functioning extremely well since then, giving proof of its essential role in the physics results achieved by the DAMPE Collaboration [3–5].

The DAMPE STK (Fig. 2) consists of 6 tracking planes, for 6 independent measurements of the position (x, y) of a charged particle traversing the STK. Each tracking plane consists of two layers of single-sided Silicon micro-Strip Detectors (SSDs) measuring the two orthogonal views perpendicular to the pointing direction of the detector. A layer is a matrix of  $8 \times 2$  ladders, each ladder (Fig. 3) consists of 4 single-sided AC-coupled SSDs, daisy-chained via micro-wire bonds. The SSDs, produced by Hamamatsu Photonics, have a thickness of  $320 \mu\text{m}$  and a sensitive area of  $9.29 \text{ cm} \times 9.29 \text{ cm}$  segmented into 768 strips with a pitch of  $121 \mu\text{m}$ . To limit power consumption, electronics density and transferred data, only one every second strip is read out, therefore the STK has 73 728 readout channels. The 384 channels of each ladder are read out by 6 VA140 ASIC chips made by IDEAS. The 192 ladders are read out in groups of 24 by 8 data acquisition Tracker Readout Boards (TRBs) located on the sides. The tracking layers are mounted on 7 support trays, the top and the bottom trays are equipped only on the side facing the five internal trays which are equipped on both sides. A support tray consists of a core of aluminium honeycomb sandwiched between two sheets of Carbon Fibre Reinforced Polymer (CFRP). To favour the photon conversion into electron–positron pairs, in the second, third and fourth tray, 1 mm thick tungsten plates are glued on the lower CFRP sheet, giving a total thickness equivalent to about one radiation length. The total power consumption of the STK is about 90 W.



**Figure 2:** Left: Exploded view of the STK. Each tracking layer is made of 64 silicon micro-strip detectors (SSDs). On each side of the STK there are two data acquisition boards (TRBs) and a radiator which dissipates the heat produced by the electronics. Right: Photograph of the STK before the assembly of the last support tray.

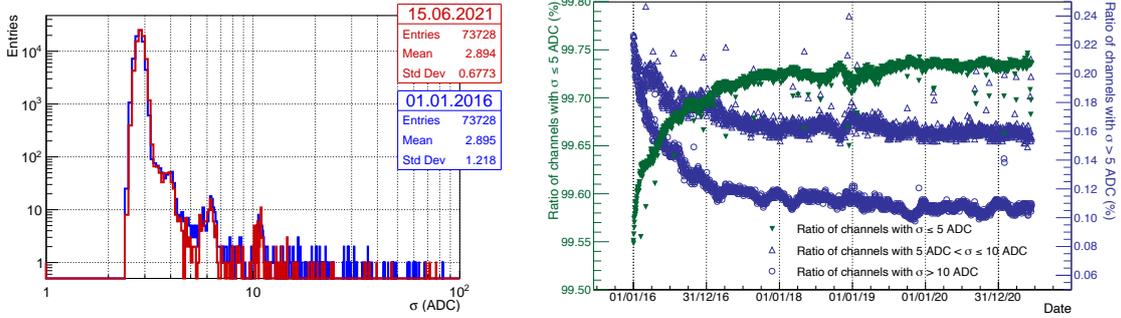


**Figure 3:** Photograph of one ladder. The 4 SSDs are daisy-chained via micro-wire bonds. The front-end electronics hosts 6 VA140 ASICs for the readout of 384 channels and 2 temperature probes.

### 3. The STK in-orbit performance

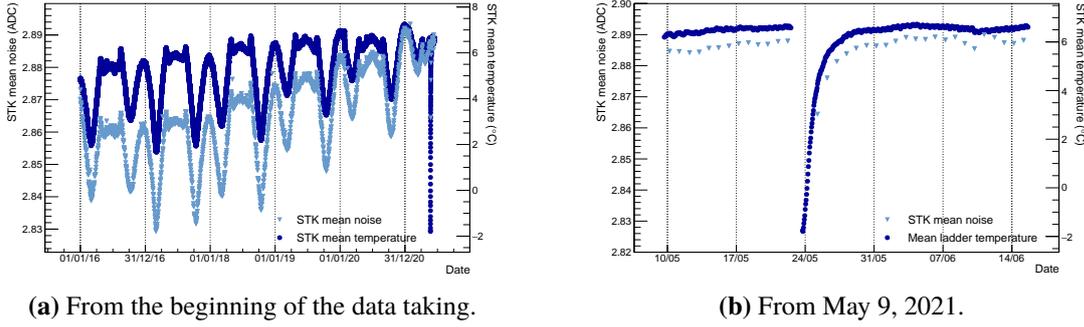
#### 3.1 Noise and temperature behaviour

The STK has always revealed an excellent noise stability. The left panel of Fig. 4 shows the distribution of the channel noise at the beginning of the mission (January 1, 2016) and recently (June 15, 2021). The two distributions have a mean  $< 3$  ADC, coincident with their most probable value, and they are very similar, denoting the excellent STK noise stability. Since the signal for a Minimum Ionizing Particle (MIP) hitting a readout strip is  $\sim 52$  ADC, the achieved signal/noise ratio is  $\sim 18$ . Another indicator of the excellent quality of the STK is the fraction of good channels, defined conservatively as the channels with a noise  $\sigma \leq 5$  ADC. The right panel of Fig. 4 shows the time evolution of the good channel fraction together with the up to date time evolution of the ratio of channels with a noise  $\sigma \in (5, 10]$  ADC and  $\sigma > 10$  ADC. At the beginning of the data acquisition, the 99.55% of the channels was good, this fraction increased over time thanks to the stabilization in space and since two years its value is stable around 99.74%.



**Figure 4:** Right: Noise distribution of the 73 728 STK readout channels at the beginning of the mission (January 1, 2016) and recently (June 15, 2021). Left: Time evolution of the ratio of channels with noise  $\sigma \leq 5$  ADC (green full triangles),  $\sigma \in (5, 10]$  ADC (blue empty triangles) and  $\sigma > 10$  ADC (blue empty circles).

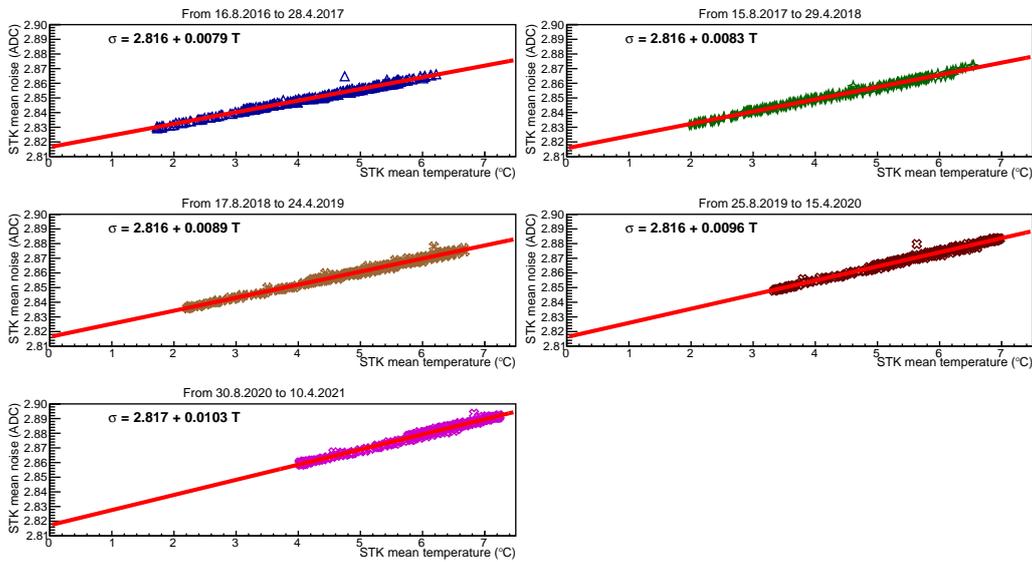
The excellent STK noise stability is correlated to the high STK temperature stability. The latter is due to the robustness of the mechanical design and the Sun-synchronous orbit. Fig. 5 shows the time evolution of the STK mean noise and the STK mean temperature. The STK mean noise is computed as the mean of all the channel noises, while the STK mean temperature is the mean of all the temperatures measured by two probes placed on the front-end electronics of each ladder. The STK mean temperature is very stable, with a daily variation much lower than  $0.1$  °C. The evident periodicity is due to the seasonal variation of the satellite orbit, moderated by the Earth's shadow from May to July. As a result the STK mean noise is very stable with a maximum variation of  $0.04$  ADC. For two days, from May 23 to 24, 2021 the payload was not powered. In the right panel of Fig. 5 we can see that only six days after the detector was powered on again, both the noise and the temperature were back at the expected levels. The correlation between the noise and the temperature can be quantified performing the linear fits shown in Fig. 6. The STK mean noise is increasing slightly more than the STK mean temperature, from  $0.008$  ADC/°C to  $0.01$  ADC/°C in five years, possibly because of a still negligible radiation damage of the silicon detectors.



(a) From the beginning of the data taking.

(b) From May 9, 2021.

**Figure 5:** Up to date time evolution of the STK mean noise (light blue triangles) and of the STK mean temperature (dark blue dots).

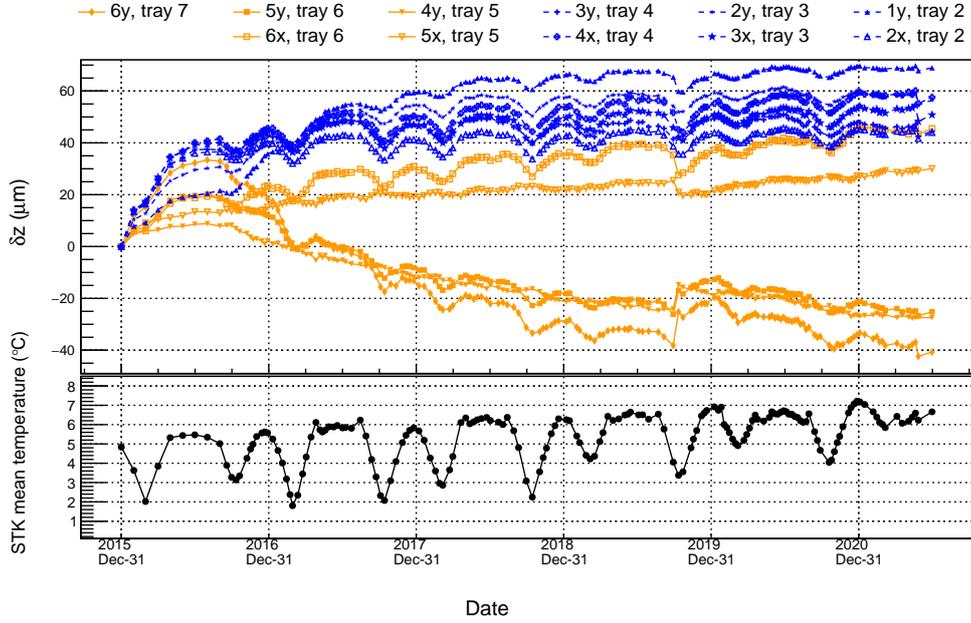


**Figure 6:** STK mean noise as a function of the STK mean temperature from August to April for every year, excluding the period when the satellite is in the Earth's shadow.

### 3.2 Alignment and position resolution

The position resolution of the silicon sensors, achieved with the analog readout that takes into account the charge sharing on the non-readout strips, is less than  $50 \mu\text{m}$  for incidence angles lower than  $40^\circ$ , and much lower than the  $70 \mu\text{m}$  achievable with a digital position finding algorithm below  $70^\circ$  [2]. The good temperature stability also ensures a good mechanical stability of the STK. The mechanical assembly of the STK has a construction precision of about  $100 \mu\text{m}$  that is larger than the position resolution of the silicon sensors, therefore an alignment procedure is needed to correct for the displacement and rotation of each sensor with respect to its nominal position, allowing the full tracker potential to be exploited [6]. Five alignment parameters are considered for each silicon sensor:  $\Delta_x/\Delta_y$ ,  $\Delta_z$ ,  $\theta_x$ ,  $\theta_y$ ,  $\theta_z$ . They correspond to the two offsets and three rotation angles of the silicon sensor, for a total of 3 840 alignment parameters. As an example, Fig. 7 shows the time evolution of the variation ( $\delta z$ ) of the average z position of the sensors of each STK tracking layer with respect to the average z position of the sensors of the first layer (1x, tray 1). The change of

this variation with time can be explained by two main effects: humidity release and temperature variation. We observe in the beginning of in-orbit operations the contraction of the support trays, due to the humidity release process, then a correlation with the temperature. The importance of the

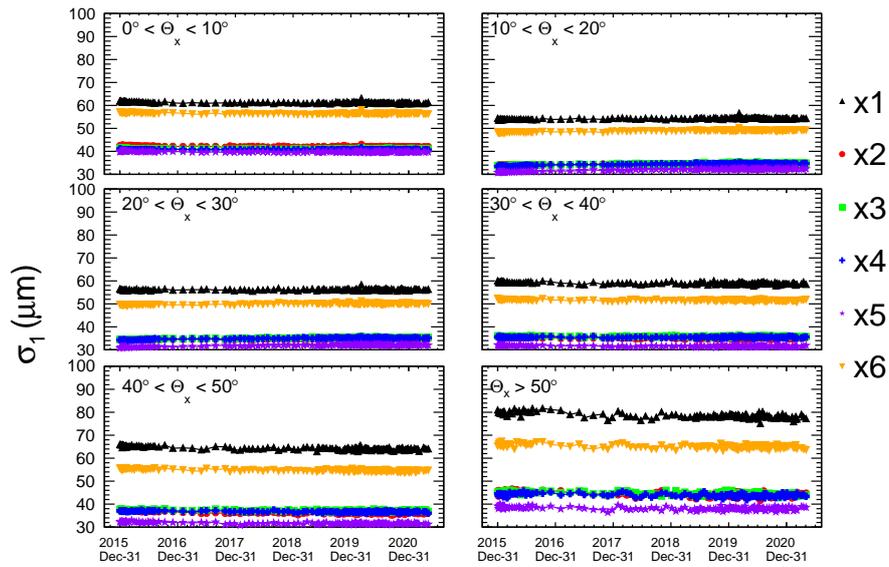


**Figure 7:** The variation of the average  $z$  position of each STK tracking layer with respect to the first layer (1x, tray 1) as a function of time (top), and the average temperature of the STK, measured by the temperature probes of all the ladders (bottom). A realignment is done once every two weeks. The dashed/blue curves correspond to the trays hosting tungsten plates.

alignment, performed once every two weeks, is demonstrated in Fig. 8, where the optimal position resolution of the 6 STK  $x$  layers, as a function of the time is shown for different incidence angles of the reconstructed tracks. After the alignment, the optimal position resolution for incidence angles below  $50^\circ$  is much lower than  $70 \mu\text{m}$ . Fig. 9 shows the variation of the optimal position resolution as a function of time. Thanks to the bi-weekly updates of the alignment parameters, the optimal position resolution remains stable and below  $6\%$  for all STK layers and particle incidence angles.

#### 4. Conclusions

The Silicon-Tungsten tracKer-converter (STK) of the DAMPE mission, based on the robust technology of single-sided silicon strip detectors with analog readout, plays a crucial role in the reconstruction of charged particle tracks, in the gamma-ray detection, and in the cosmic ray charge measurement. The STK is showing an excellent noise, thermal and mechanical stability since the beginning of the data taking. About the totality of channels ( $99.74\%$ ) are superbly performing, with a noise less than 5 ADC. In-orbit calibration and alignment procedures allow to achieve the best possible position and angular resolutions and long term stability.



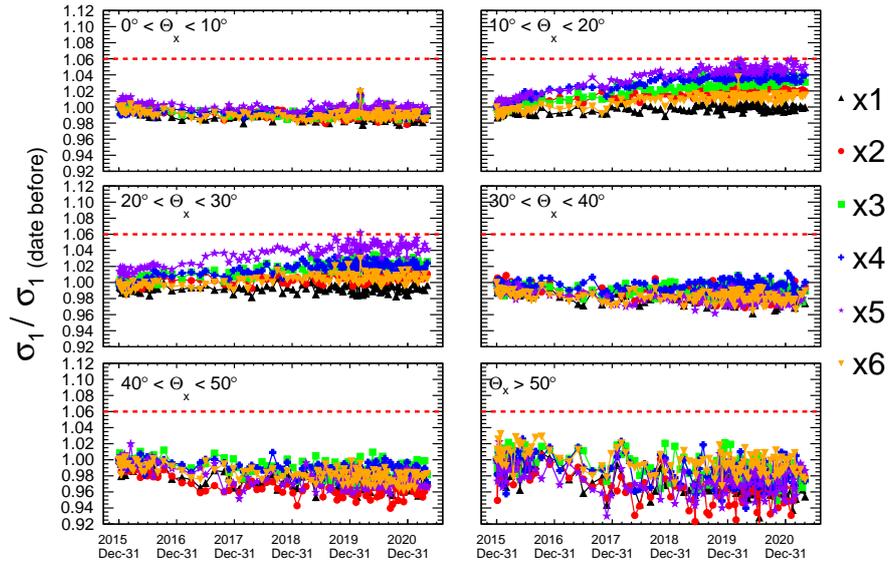
**Figure 8:** Optimal position resolution for the six x layers at different track inclinations as a function of time. The values are shown for x layers only, while y layers show a similar behaviour.

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**Figure 9:** The variation of the optimal position resolution for the six x layers at different track inclinations as a function of time. The values are shown for x layers only, while y layers show similar behaviour. Horizontal dashed lines are shown to indicate the 6% deviation from the initial values. The alignment is performed once every two weeks.

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