

The 23 July 2012 SEP event numerical simulation with multi-spacecraft observation data

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23 July 2012, multiple-spacecraft, namely STEREO-A, STEREO-B, and ACE, observed an extremely powerful, superfast interplanetary coronal mass ejection (ICME) together with the ICME-driven shock and associated solar energetic particles (SEPs). We analyze the relationship between the propagation of the shock and the SEP flux with the Parker spiral magnetic field model. Moreover, we simulate the SEP event by numerically solving the three-dimensional focused transport equation of SEPs considering the shock as the moving source of energetic particles. We use the same diffusion model format for the simulations of protons and electrons but with different parameters for simplicity. The simulation results can qualitatively explain the important features of the SEP flux observed by the multiple spacecraft simultaneously. Additionally, the numerical results for both energetic protons and electrons approximately agree with multi-spacecraft observations.

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1. Introduction

On 23 July 2012, a halo superfast ICME was launched from the Sun with a maximum speed reaching $3050 \pm 260 \text{ km s}^{-1}$ with the magnetic field of the ejecta reaching $109 \pm 1 \text{ nT}$ [1]. Significant SEPs assumed to be accelerated by the ICME-driven interplanetary shock were detected during this event [2].

Here, we investigate the extreme 23 July 2012 SEP event observed by multi-spacecraft *STEREO-A*, *STEREO-B*, and *ACE*. We apply a simple and ideal Parker spiral magnetic field model to analyze the magnetic connections of the shock to the multiple spacecraft with the outward propagating CME. Furthermore, we study SEP flux by numerical simulations using the parameters observed by the spacecraft, and we compare our results with the observations. We use the numerical Shock Particle Transport Code (SPTC) by [3] which considers an ICME shock to be a moving source of SEPs. In section 2, we present the observations and analysis. In section 3, we show the transport model. In section 4, we show our simulations and the comparisons with observations. In section 5, we present the conclusion and discussion.

2. Observations and analysis

The 23 July 2012 CME was observed launching from the Sun by the EUVI observations of the SECCHI instrument onboard the two *STEREO* spacecraft at about 0208 ± 2 minutes UT [2]. The source location was the sunspot group NOAA 11520 at $S15^\circ W133^\circ$ [1]. The fast forward shock driven by the CME was observed to reach *STEREO-A* at 20:55:25 UT (T_{STA}). At 22:55 UT, the leading edge of the ICME reached *STEREO-A* in the ecliptic with latitude and longitude $S0.07^\circ W121.3^\circ$, and heliocentric distance 0.96 au [2]. The latitude and longitude of *STEREO-B* is $S0.16^\circ E115.2^\circ$. Later, the shock reached *STEREO-B* at 21:21:01 UT (T_{STB}), but there was no shock encounter on *Advanced Composition Explorer (ACE)* which was 121° east of *STEREO-A*.

In order to determine the magnetic connections between the multiple spacecraft together with the shock front, we use a schematic (Figure 1) which shows their relative positions. We use the Parker spiral model to describe the magnetic field for simplicity. Figure 1 shows such a schematic in the ecliptic plane at 20:55 UT on 23 July, 2012. The red and blue spirals depict Parker spiral magnetic field lines passing through *STEREO-A* and *STEREO-B* at $W121.3$, 0.96 au, and $E115.2$, 1.02 au, respectively. The black spiral represents a Parker magnetic field line passing through *ACE* at $W0$, 1 au. The black straight line points to the direction of the shock nose. The yellow area indicates the scope of the shock sweep with width W_s , the possible value of which can be deduced based on spacecraft observations.

The time profiles of proton fluxes observed by the multiple spacecraft during the 23 July 2012 SEP event are exhibited in Figure 2. The above proton channels with similar energy are chosen for comparison. In order to investigate the transport of the CME-driven shock and its contributions to the SEP fluxes detected by the multiple spacecraft, we divide the proton flux time profile into several parts by time with various vertical lines according to some key moments in the propagation of the shock. The dotted vertical line indicates $T1$, 14:08:00 UT on July 23, 2012, when the shock connection for *ACE* was lost. The red and blue vertical solid lines indicate the shock arrival time, 20:55 UT at *STEREO-A* and 21:21 UT at *STEREO-B*, respectively, which almost coincide in Figure

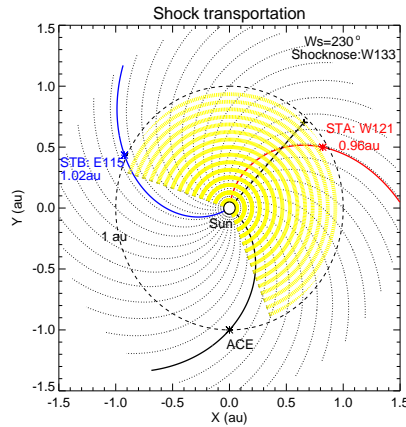


Figure 1: Positions of the spacecraft and the shock nose direction in the ecliptic plane at 20:55 UT on 23 July 2012, the time when the shock passed *STEREO-A*.

2. The dashed vertical lines indicate T_2 , 15:19:59 UT on July 25, when the connection of *ACE* was reestablished. In addition, the dot-dashed vertical lines indicate T_3 , 06:56:00 UT on July 27, when the connection of *STEREO-B* was lost again. Similarly to Figure 2, the electron flux time profile is shown in Figure 3.

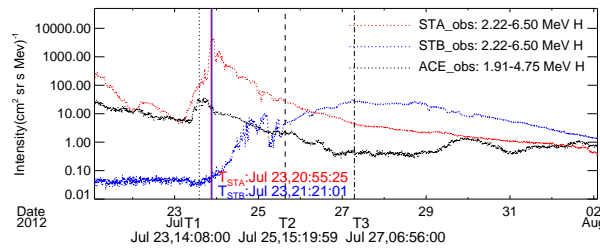


Figure 2: Time profile of proton fluxes by the multi-spacecraft observation during the 23 July 2012 SEP event.

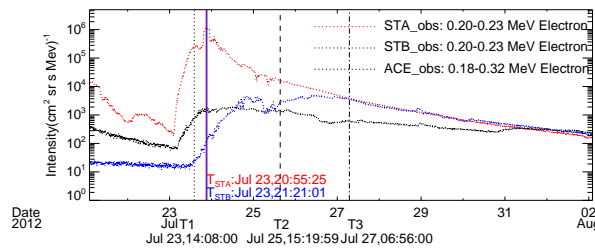


Figure 3: Time profile of electron fluxes by the multi-spacecraft observation during the 23 July 2012 SEP event.

At T_1 , as the cobpoint of *STEREO-A*, which indicates the location of the shock front that is magnetically connected to the observer [3, 5], is near the shock nose with the acceleration efficiency much higher than at the shock edge, the SEP flux detected by *STEREO-A* (the red curves in Figure

2 for protons and Figure 3 for electrons) was much higher than the background level. In addition, when the shock reached *STEREO-A* at 0.96 au, T_{STA} , the SEP flux of *STEREO-A* almost reached the peak. We note that during the shock crossing there was a significant increase of the SEP flux observations by *STEREO-A*. Soon after, shock reached *STEREO-B* at 1.02 au at T_{STB} . Afterward, as the shock passed *STEREO-A*, the SEP flux detected by *STEREO-A* dropped dramatically.

Solar energetic particle flux detected by *STEREO-B* remained at background level before T_1 because the shock was not wide enough to cover *STEREO-B*'s magnetic field line (Figure 4). However, as connection to the shock was established some time after T_1 , the cobpoint of *STEREO-B* was moving toward the shock nose and the SEP flux observed by *STEREO-B* rose with time. We note that the SEP flux observation of *STEREO-B* began to rise around T_1 before the connection was actually established, one may assume that this is due to perpendicular diffusion of the energetic particles.

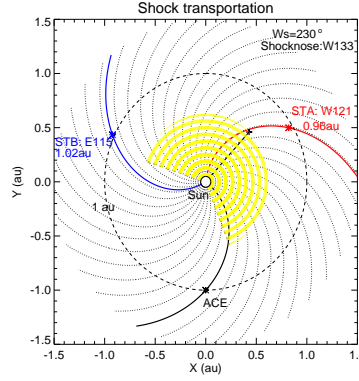


Figure 4: Same as Figure 1 except that the shock was at around 0.6 au on T_1 , July 23, 14:08:00.

Around T_1 , *ACE* observed an SEP intensity maximum for protons, and after about 8 hours SEP intensity started to decrease (Figure 2). However, for electrons, around T_1 , *ACE* observed SEP intensity to continuously increase with time until it reached its maximum in a few hours and began to decrease after a few days (Figure 3). Figure 4 shows that at T_1 the shock arrived at the position around 0.63 au. We can see that the cobpoint of *ACE* moved out from the edge of the shock, and so there was no connection between *ACE* and the shock.

3. Solar energetic particle transport model

In this work, we use the shock particle transport code [3, 8] later abbreviated to SPTC [10] to simulate the transport of the 23 July 2012 SEP event assuming the CME-driven shock to be a moving particle source. For the study of the transport mechanism of the SEPs [11, 12], we use the three-dimensional focused transport equation [11–14]:

$$\begin{aligned}
 & \frac{\partial f}{\partial t} + \left(v\mu\hat{\mathbf{b}} + \mathbf{V}^{sw} \right) \cdot \nabla f - \nabla \cdot \left(\kappa_{\perp} \cdot \nabla f \right) - \frac{\partial}{\partial \mu} \left(D_{\mu\mu} \frac{\partial f}{\partial \mu} \right) \\
 & - p \left[\frac{1 - \mu^2}{2} \left(\nabla \cdot \mathbf{V}^{sw} - \hat{\mathbf{b}}\hat{\mathbf{b}} : \nabla \mathbf{V}^{sw} \right) + \mu^2 \hat{\mathbf{b}}\hat{\mathbf{b}} : \nabla \mathbf{V}^{sw} \right] \frac{\partial f}{\partial p} \\
 & + \frac{1 - \mu^2}{2} \left[-\frac{v}{L} + \mu \left(\nabla \cdot \mathbf{V}^{sw} - 3\hat{\mathbf{b}}\hat{\mathbf{b}} : \nabla \mathbf{V}^{sw} \right) \right] \frac{\partial f}{\partial \mu} = 0,
 \end{aligned} \tag{1}$$

where $D_{\mu\mu}$ is the particle pitch-angle diffusion coefficient following [15], [16], and [17]. A quasi-linear model with the nonlinear effect of magnetic turbulence on the pitch angle scattering at $\mu = 0$ [18] can be written as

$$D_{\mu\mu}(\mu) = \frac{D_{slab}v}{l_{slab}} \left(\frac{R}{R_{au}} \right)^{-1/3} (|\mu|^{g-1} + h)(1 - \mu^2). \quad (2)$$

The relationship between the particle pitch angle diffusion coefficient and the parallel particle mean free path is [15, 19, 20]

$$\lambda_{\parallel} = \frac{3}{8}v \int_{-1}^1 \frac{(1 - \mu^2)^2}{D_{\mu\mu}} d\mu, \quad (3)$$

and the parallel diffusion coefficient κ_{\parallel} can be written as $\kappa_{\parallel} = v\lambda_{\parallel}/3$.

[21] developed nonlinear guiding center (NLGC) theory for perpendicular diffusion, which was approximated in analytic form [22] for particles in certain parameter and energy ranges,

$$\kappa_{\perp} = \frac{1}{3}vD_{2D}l_{2D}^{2/3} \times \lambda_{\parallel}^{1/3} (\mathbf{I} - \hat{\mathbf{b}}\hat{\mathbf{b}}), \quad (4)$$

Here, D_{2D} is a parameter that depends on the spectral index in the inertial range and the 2D component of the magnetic turbulence.

Our model assumes the shock to be the source of particle injection with the boundary condition [23]:

$$f_b = a\delta(r - v_s t) \left(\frac{r}{r_c} \right)^{\alpha} \exp \left[-\frac{|\phi(\theta, \varphi)|}{\phi_c(p)} \right] p^{\gamma} \xi(\theta, \varphi), \quad (5)$$

where the shock acceleration efficiency parameter α measures the damping rate with radial distance, and the other shock acceleration strength parameter ϕ_c describes the injection decrease from the middle to the flank of the shock. Here, γ is the power-law spectrum index of the shock, and $\xi(\theta, \phi)$ shows the angular range of the shock front [3],

$$\xi(\theta, \varphi) = \begin{cases} 1 & \text{if } |\phi(\theta, \varphi)| \leq \phi_s \\ 0 & \text{otherwise,} \end{cases} \quad (6)$$

where $\phi(\theta, \varphi)$ is the angle between any particle injection position at the shock front and shock nose, and ϕ_s is the half angular width of the shock.

We reformulate the transport equation (1) in terms of a set of stochastic differential equations, and solve it with a time-backward Markov stochastic process method using the Monte Carlo simulation [24]. The particles are traced from the observation time back to the initial time of the injection at the source [11].

4. Simulations and comparisons with observations

From *STEREO*/SEPT we choose the proton channel with the energy range 2.2 – 6.5 MeV, and from *ACE*/EPAM we choose the proton channel with the energy range 1.91 – 4.75 MeV. We use 3 MeV as the typical value for simulations to compare with the three observational proton channels.

For each simulated data point of energetic particle flux, 2.88×10^7 pseudo-particles are used. The observation and simulation results of the time profile of SEP fluxes are shown in Figure 5. The dotted and solid lines indicate observations and simulations, respectively. The black, red, and blue lines correspond to *ACE*, *STEREO-A*, and *STEREO-B*, respectively.

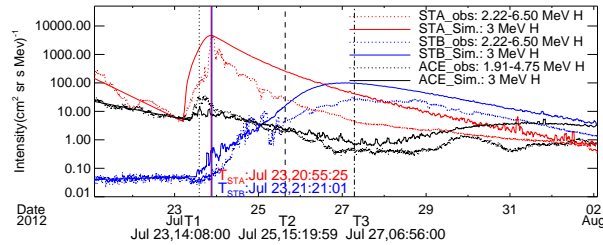


Figure 5: Proton fluxes of the observations and simulations during the 23 July 2012 SEP event.

From Figure 5 we can see that the simulations and observations generally agree well for the three spacecraft. Particularly in terms of the timing for the start and peak of SEP flux, the simulations approximately agree with observations.

We also perform numerical simulations for 0.2 MeV electrons with 1.091×10^7 pseudo-particles for each data point. In Figure 6 we compare the simulation results with the electron observations of the EPAM instrument onboard *ACE* and the SEPT instruments onboard *STEREO-A* and *STEREO-B*. The ranges of energy we choose for the electron channel are 0.18 – 0.32 MeV and 0.20 – 0.23 MeV for EPAM and SEPT, respectively. From Figure 6 we can see that the observations and simulations of *STEREO-A* are consistent except that the peak of observations is about one order of magnitude higher. On the other hand, the *STEREO-B* observations, with a later starting time than simulations, have a peak that is lower than the simulated one. Furthermore, the *ACE* observations and simulations agree relatively well. However, around *T3*, there is an increase in the simulation of the flux as would be observed by *ACE*. In SEP events, reservoir phenomena are usually observed during which particle intensities are nearly the same at different locations in decay phase[e.g., 6–8].

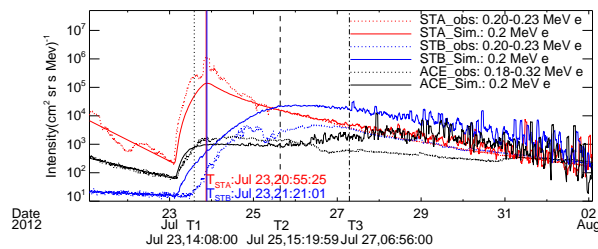


Figure 6: Electron fluxes of the observations and simulations during the 23 July 2012 SEP event.

5. Conclusion and Discussion

In this work, we investigate the 23 July 2012 SEP event observed by multi-spacecraft *STEREO-A*, *STEREO-B*, and *ACE*.

First, we used the Parker's spiral magnetic field model to qualitatively analyze the relationship between the propagation of the CME-driven shock in the interplanetary space and the associated SEP flux observed by the multiple spacecraft. We provide a schematic to show their relative positional relationship. Our analysis is able to qualitatively explain some of the important features, especially in terms of the timing for the start and peak of SEP flux observed by *STEREO-A*, *STEREO-B*, and *ACE*, simultaneously.

We then simulated the SEP event using the three-dimensional focused transport model, by treating the shock as a moving energetic particle source. In the simulations, almost all the important particle transport effects, such as solar wind convection, particle streaming along the magnetic field line, magnetic focusing, adiabatic cooling, and the diffusion coefficients parallel and perpendicular to the IMF, are included. The simulations and observations approximately agree for the three spacecraft, especially in terms of the timing for the start and peak of SEP flux. Further, the schematic can qualitatively describe some characteristics of simulations and observations.

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